AIRCRAFT WEIGHING REPORT

MODEL:	SERIAL NUMBER:	REGISTRATION NUMBER:	DATE:
AT 01	159	F-HARR	4/11/11

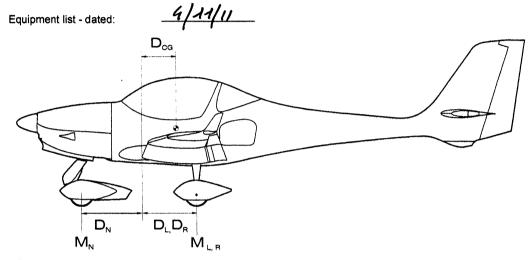
Data in accordance with Airplane Flight Manual

Reference Datum:

Horizontal reference line: Weighing Conditions:

Leading edge of wing root rib. Refer to 08-20-00 for information on longitudinal leveling.

Including brake fluid, eng. oil and coolant, unusable fuel.



Position	Gross ~ kg (lbs)	Tare ~ kg (lbs)	Net Weight ~ kg (I	bs) Lever Arm ~ m (in.)
Nose Wheel			M _N = 113	D _N = - 0,85
Right Main Wheel			M. = 136	D. = + 0,779
Left Main Wheel		The state of the s	$M_{R} = 196$	DR = +0, 779

Empty Weight Moment:

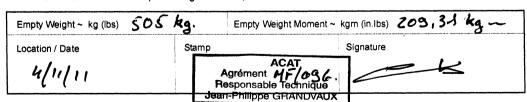
$$MO_{Empty} = M_N \times D_N + M_L \times D_L + M_R \times D_R = 209,318$$
 kgm (in.lbs)
 $D_{SL} = MO_{Empty} / M_{Empty} = 0,414$ m (in.)

CG Position for Empty Weight:

$$D_{sL} = MO_{Empty} / M_{Empty} = O_{sL} + MG \qquad m (in.)$$

Maximum Useful Load	+ MTOW Empty Weight	+	750 505
·	= Max. Useful Load		245

Data to be entered into the Airplane Flight Manual, Section 6:



Aircraft Weighing Form Figure 201